

Solutions of $\left(v_r - \frac{v_c^2}{v_r}\right) \frac{\partial v_r}{\partial r} = \frac{2 v_c^2}{r} - \frac{G M_\odot}{r^2}$

$$\frac{\partial v_r}{\partial r} \Rightarrow \frac{d v_r}{d r}$$

integrate with r

$$\int \left(v_r - \frac{v_c^2}{v_r}\right) d v_r = \int \left(\frac{2 v_c^2}{r} - \frac{G M_\odot}{r^2}\right) d r$$

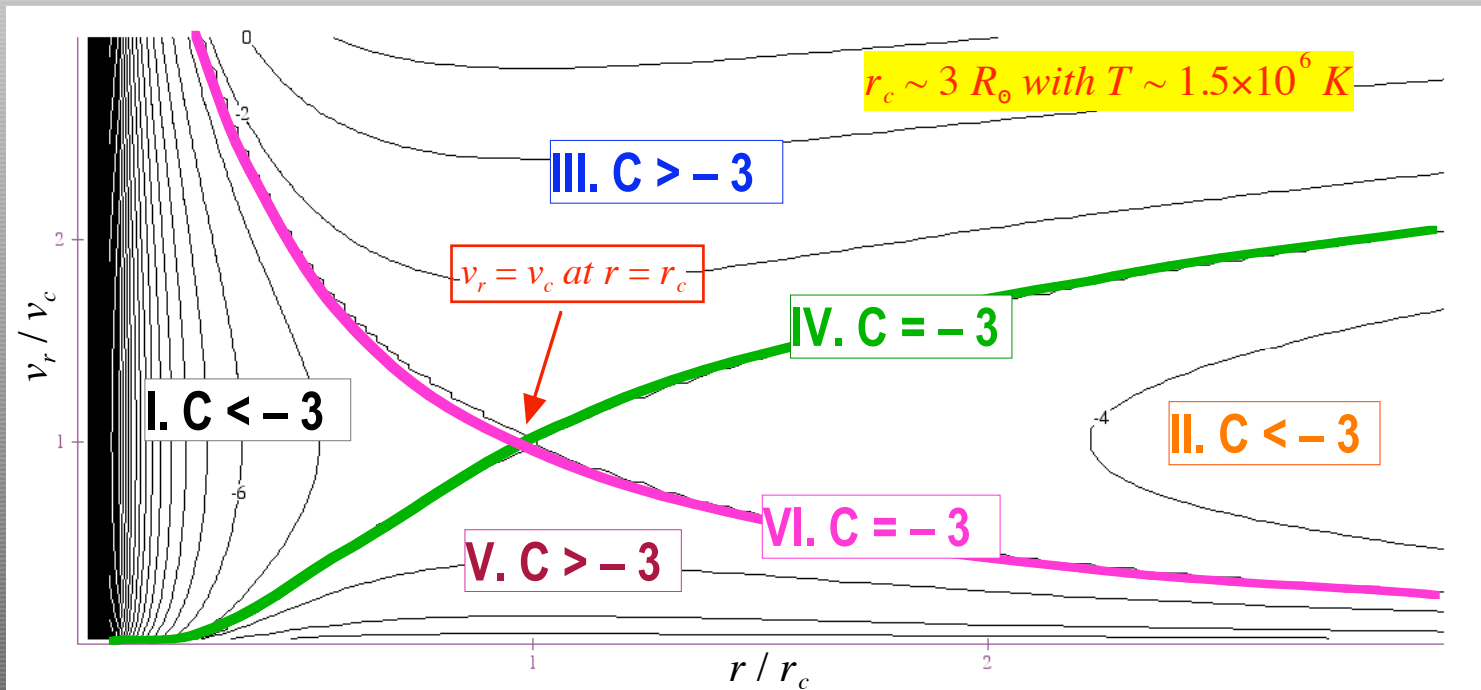
dimensionless form

$$\left(\frac{v_r}{v_c}\right)^2 - \ln \left(\frac{v_r}{v_c}\right)^2 = 4 \ln \frac{r}{r_c} + 4 \left(\frac{r}{r_c}\right)^{-1} + C$$

$$r_c \equiv \frac{G M_\odot}{2 v_c^2}: \text{critical radius}$$

$C = \text{constant}$

$$C = -3 \text{ when } (r, v_r) = (r_c, v_c).$$

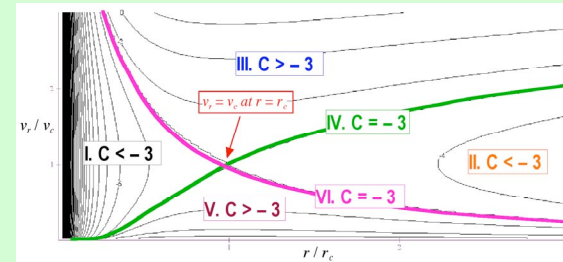


Properties of the solutions...

I. $C < -3$

II. $C < -3$

These two solutions are **physically incorrect** (although they are correct mathematically) because flow velocity takes two different values at the same position.



III. $C > -3$

IV. $C = -3$

V. $C > -3$

VI. $C = -3$

These four solutions are physically correct. Which is appropriate for solar wind?

III. $C > -3$

VI. $C = -3$

... starts with a **supersonic flow** ($v_r > v_c$) at a solar surface

=> inconsistent with inner boundary condition

IV. $C = -3$

... starts with a **subsonic flow** ($v_r < v_c$) at a solar surface, and becomes **supersonic**

=> consistent with inner boundary condition

V. $C > -3$

... starts with a **subsonic flow** ($v_r < v_c$) at a solar surface, and remains **subsonic**

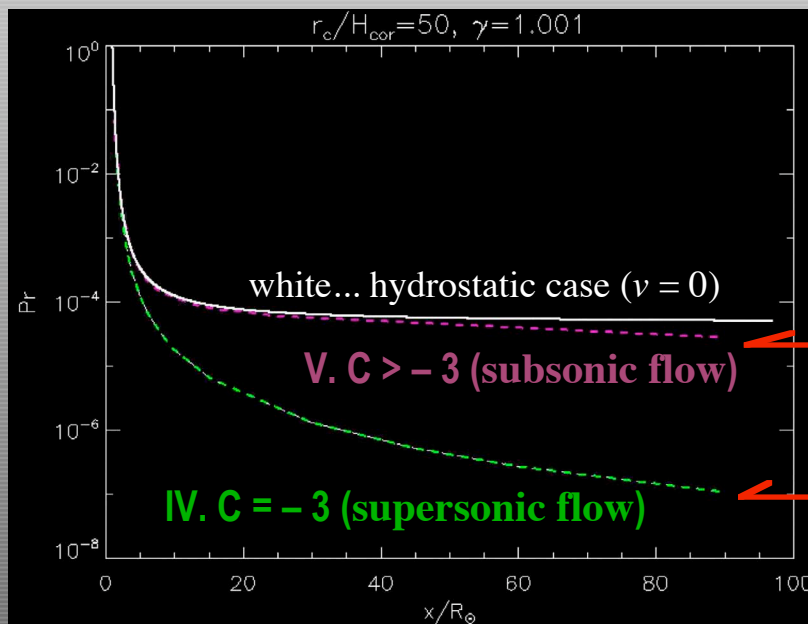
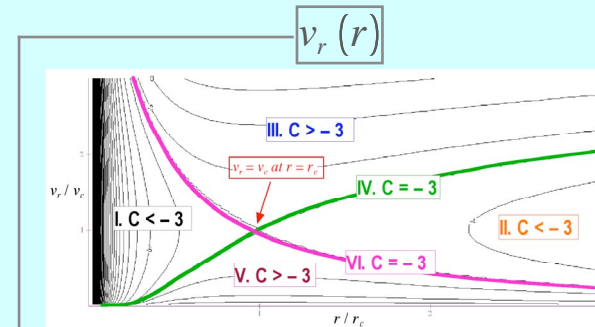
=> consistent with inner boundary condition

Outer boundary condition (solution IV vs. solution V)...

We focus on **gas pressure distribution**:

$$p(r) = \rho(r) \frac{k_B T}{\bar{m}} = \rho(r) v_c^2, \quad (v_c \text{ is constant})$$

$$\rho(r) = \frac{\rho(R_0) v_r(R_0) R_0^2}{v_r(r) r^2}$$



The value of gas pressure at outer boundary is different between IV (supersonic case, low pressure) and V (subsonic case, high pressure).



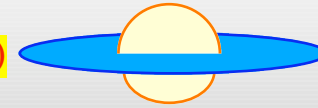
Observation suggests a very low value of gas pressure at the outer boundary ($P^{interstellar} \sim 10^{-17} P^{solar\ surface}$), so we select **solution IV (supersonic case)**.

Magnetohydrodynamic model

Weber & Davis 1.5-dimensional model

r & ϕ -components of a vector

(axisymmetric, on the equatorial plane, polytropic, steady, rotation)
 (depends on r , r & ϕ -components of a vector are considered)



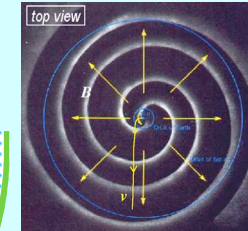
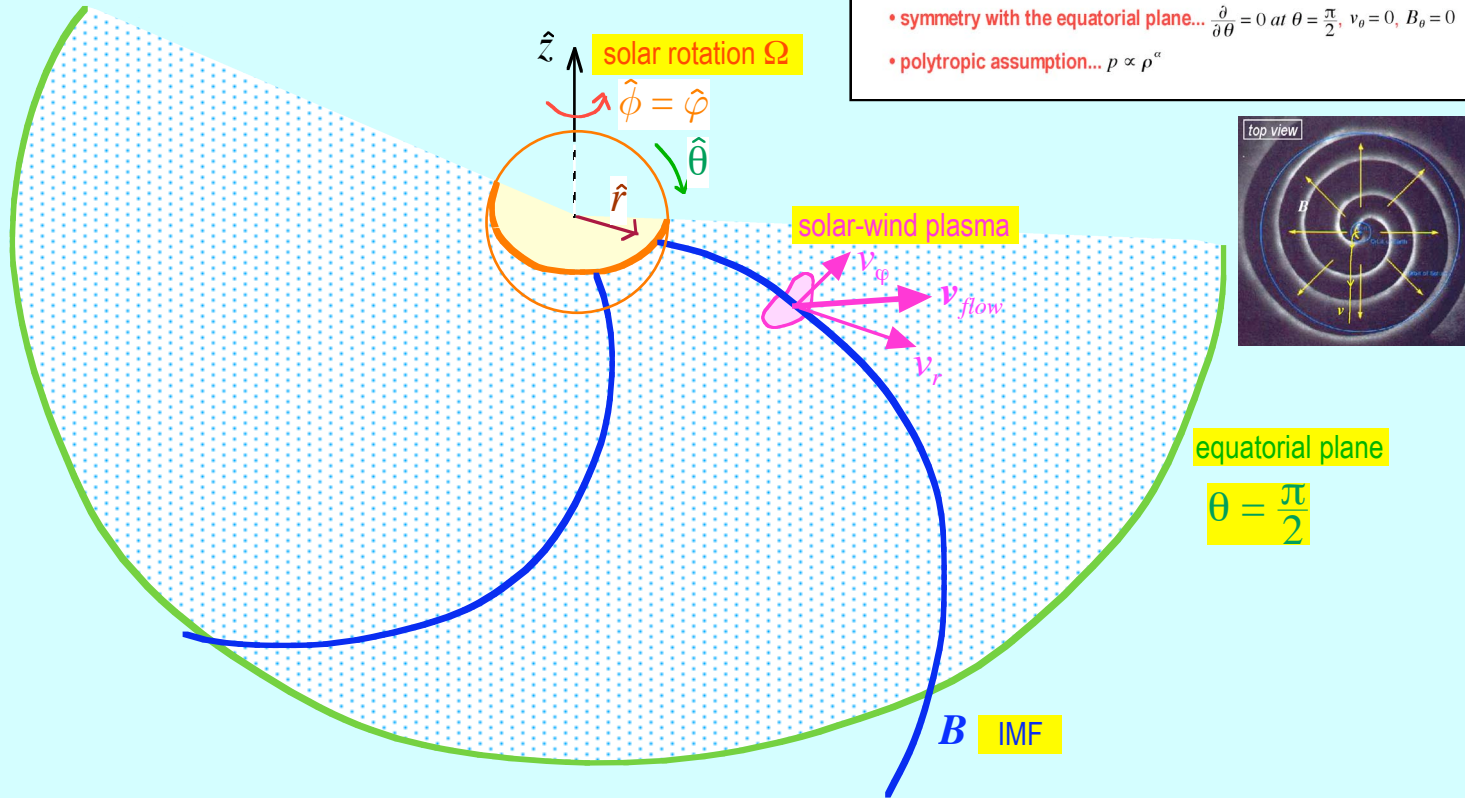
(r, θ, ϕ) ... spherical coordinates with $\theta = \pi/2$

$$\mathbf{v}(r, \theta, \phi) = v_r(r, \theta, \phi) \hat{r} + v_\theta(r, \theta, \phi) \hat{\theta} + v_\phi(r, \theta, \phi) \hat{\phi}$$

$$\rightarrow v_r\left(r, \theta = \frac{\pi}{2}\right) \hat{r} + v_\phi\left(r, \theta = \frac{\pi}{2}\right) \hat{\phi}$$

Assumption:

- axial symmetry... $\frac{\partial}{\partial \phi} = 0$
- steady state... $\frac{\partial}{\partial t} = 0$
- only equatorial plane is considered... $\theta = \frac{\pi}{2}$
- symmetry with the equatorial plane... $\frac{\partial}{\partial \theta} = 0$ at $\theta = \frac{\pi}{2}$, $v_\theta = 0$, $B_\theta = 0$
- polytropic assumption... $p \propto \rho^\alpha$



Basic equations (differential form):

CGS unit $\frac{\partial}{\partial r} \Rightarrow \frac{d}{dr}$

mass conservation...

$$\nabla \cdot (\rho \mathbf{v}) = 0 \quad \longrightarrow \quad \frac{1}{r^2} \frac{d}{dr} (r^2 \rho v_r) = 0$$

momentum equation...

$$\rho \mathbf{v} \cdot \nabla \mathbf{v} = -\nabla p + \frac{1}{4\pi} (\nabla \times \mathbf{B}) \times \mathbf{B} - \frac{GM_\odot \rho}{r^2} \hat{\mathbf{r}} = 0$$

$$\begin{array}{l} \text{r-component} \quad \rho v_r \frac{dv_r}{dr} - \frac{\rho v_\varphi^2}{r} = -\frac{dp}{dr} - \frac{1}{4\pi} \frac{B_\varphi}{r} \frac{d}{dr} (r B_\varphi) - \frac{GM_\odot \rho}{r^2} \\ \text{\(\phi\)-component} \quad \rho v_r \frac{d}{dr} (r v_\varphi) = \frac{1}{4\pi} B_r \frac{d}{dr} (r B_\varphi) \end{array}$$

energy equation...

$$\mathbf{v} \cdot \nabla \left(\frac{p}{\rho^\alpha} \right) = 0 \quad \longrightarrow \quad v_r \frac{d}{dr} \left(\frac{p}{\rho^\alpha} \right) = 0$$

induction equation...

$$\nabla \times (\mathbf{v} \times \mathbf{B}) = \mathbf{0} \quad \xrightarrow{\text{\(\theta\)-component}} \quad \frac{d}{dr} (r [v_\varphi B_r - v_r B_\varphi]) = 0$$

magnetic flux conservation...

$$\nabla \cdot \mathbf{B} = 0 \quad \longrightarrow \quad \frac{1}{r^2} \frac{d}{dr} (r^2 B_r) = 0$$

Basic equations (integral form):

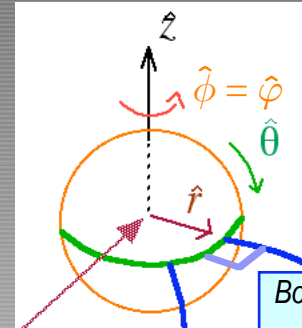
mass conservation... $r^2 \rho v_r = f$
(constant)

energy equation... $\frac{p}{\rho^\alpha} = K$
(constant)

induction equation... $v_\varphi B_r - v_r B_\varphi = \frac{C_0}{r} = \frac{\Omega R_\odot^2 B_0}{r}$

at $r = R_\odot, v_\varphi = \Omega R_\odot, B_r = B_0, B_\varphi = 0 \Rightarrow C_0 = \Omega R_\odot^2 B_0$
(constant)

magnetic flux conservation... $r^2 B_r = \Phi = R_\odot^2 B_0$
(constant)



Boundary condition:
magnetic field is normal to
solar surface

$v_\varphi B_r - v_r B_\varphi = \Omega r B_r$

$\int \phi$ -component of momentum $\times dr$

=> angular momentum conservation (per unit mass)

$$\rho v_r \frac{d}{dr} (r v_\phi) = \frac{1}{4\pi} B_r \frac{d}{dr} (r B_\phi)$$

$$\begin{aligned} \rho v_r &= \frac{f}{r^2} = \frac{f}{\Phi} = \text{const.} \\ B_r &= \frac{\Phi}{r^2} = \frac{f}{\Phi} \\ r^2 \rho v_r &= f \\ r^2 B_r &= \Phi \end{aligned}$$

integrate with r

$$r v_\phi - \frac{B_r}{4\pi \rho v_r} r B_\phi = L \dots \text{angular momentum per unit mass (constant)}$$
$$L = v_\phi(r = R_0) R_0 = \Omega R_0 R_0$$

Angular momentum conservation (per unit mass)

\int r -component of momentum $\cdot dr \Rightarrow$ energy conservation (per unit mass)

$$v_r \frac{dv_r}{dr} - \frac{v_\phi^2}{r} = - \frac{1}{\rho} \frac{dp}{dr} - \frac{1}{4\pi} \frac{B_\phi}{\rho r} \frac{d}{dr} (r B_\phi) - \frac{G M_\odot}{r^2}$$

$$\frac{p}{\rho^\alpha} = K$$

$$\frac{\alpha}{\alpha-1} \frac{d}{dr} \left(\frac{p}{\rho} \right)$$

$$\rho v_r \frac{d}{dr} (r v_\phi) = \frac{1}{4\pi} B_r \frac{d}{dr} (r B_\phi)$$

ϕ -component of momentum eq.

$$\frac{v_r}{r} \frac{B_\phi}{B_r} \frac{d}{dr} (r v_\phi)$$

$$v_\phi B_r - v_r B_\phi = \Omega r B_r \Rightarrow \frac{B_\phi}{B_r} = \frac{v_\phi - r \Omega}{v_r}$$

Induction eq.

$$\frac{v_\phi - r \Omega}{r} \frac{d}{dr} (r v_\phi) = (v_\phi - r \Omega) \frac{d}{dr} (v_\phi - r \Omega) + \frac{v_\phi^2}{r} - r \Omega^2$$

$$v_r \frac{dv_r}{dr} = - \frac{\alpha}{\alpha-1} \frac{d}{dr} \left(\frac{p}{\rho} \right) - (v_\phi - r \Omega) \frac{d}{dr} (v_\phi - r \Omega) - \frac{G M_\odot}{r^2} + r \Omega^2$$

integrate with r

radial flow kinetic energy

gravitational energy

Energy conservation (per unit mass)

$$\frac{1}{2} v_r^2 + \frac{1}{2} (v_\phi - r \Omega)^2 + \frac{\alpha}{\alpha-1} \frac{p}{\rho} - \frac{G M_\odot}{r} - \frac{1}{2} (r \Omega)^2 = E \dots \text{energy per unit mass (constant)}$$

rotational energy
thermal energy
centrifugal energy