

Basic equations:

Mass conservation.. $\rho v_r A = \text{const.}$

spherically symmetric $\Rightarrow A(r) \propto r^2$

Momentum equation...

$$\rho v_r \frac{\partial v_r}{\partial r} = - \frac{\partial p}{\partial r} - \frac{G M_\odot \rho}{r^2}$$

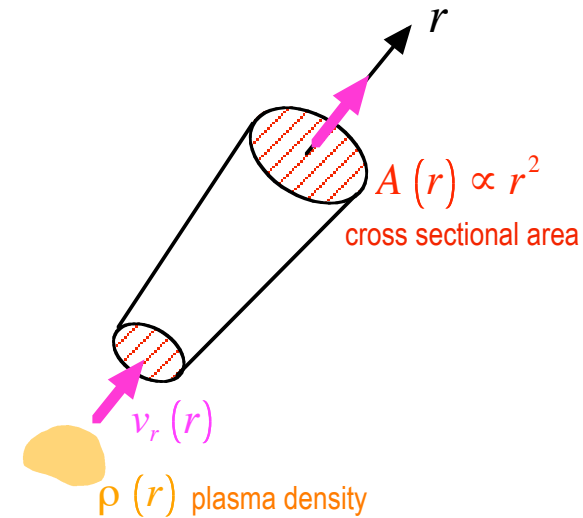
Energy equation...

$$T = \text{const.} \left(\frac{p}{\rho} = \frac{k_B T}{\bar{m}} = v_c^2 = \text{const.} \quad v_c = \sqrt{\frac{k_B T}{\bar{m}}} \right)$$

isothermal sound speed

eliminate ρ and p

$$\left(v_r - \frac{v_c^2}{v_r} \right) \frac{\partial v_r}{\partial r} = \frac{2 v_c^2}{r} - \frac{G M_\odot}{r^2}$$



Lagrangian derivative vs. Eulerian derivative

$$\frac{d}{dt} \equiv \frac{\partial}{\partial t} + (\mathbf{v} \cdot \nabla)$$

$$d A (x, y, z, t) = \frac{\partial A}{\partial x} d x + \frac{\partial A}{\partial y} d y + \frac{\partial A}{\partial z} d z + \frac{\partial A}{\partial t} d t$$

total differential

$$\begin{aligned} \therefore \frac{d A}{d t} &= \frac{\partial A}{\partial x} \frac{d x}{d t} + \frac{\partial A}{\partial y} \frac{d y}{d t} + \frac{\partial A}{\partial z} \frac{d z}{d t} + \frac{\partial A}{\partial t} \frac{d t}{d t} \\ &= \frac{\partial A}{\partial x} v_x + \frac{\partial A}{\partial y} v_y + \frac{\partial A}{\partial z} v_z + \frac{\partial A}{\partial t} \\ &= \left(v_x \frac{\partial}{\partial x} + v_y \frac{\partial}{\partial y} + v_z \frac{\partial}{\partial z} \right) A + \frac{\partial A}{\partial t} \\ &= \frac{\partial A}{\partial t} + (\mathbf{v} \cdot \nabla) A \end{aligned}$$

Solutions of $\left(v_r - \frac{v_c^2}{v_r}\right) \frac{dv_r}{dr} = \frac{2 v_c^2}{r} - \frac{G M_\odot}{r^2} \dots$

$\frac{\partial v_r}{\partial r} \Rightarrow \frac{dv_r}{dr}$

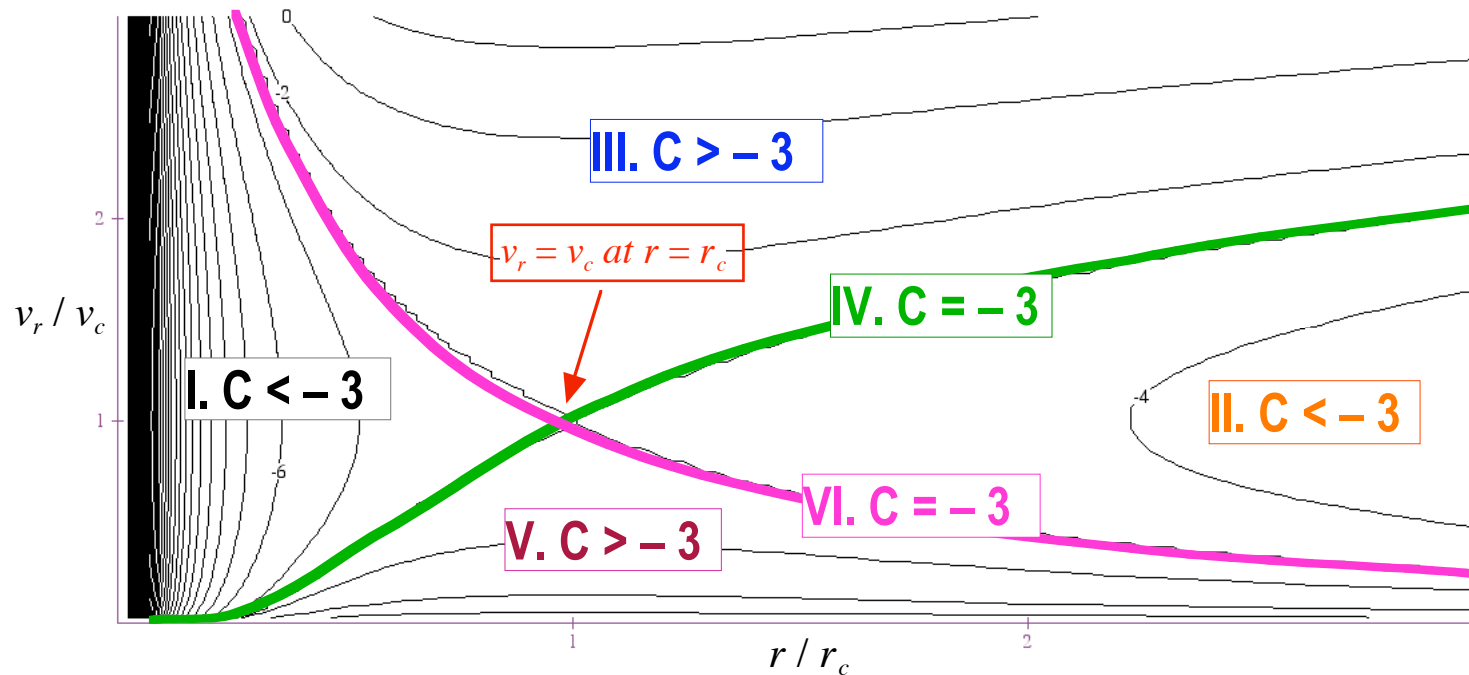
integrate

dimensionless form

$$\left(\frac{v_r}{v_c}\right)^2 - \ln \left(\frac{v_r}{v_c}\right)^2 = 4 \ln \frac{r}{r_c} + 4 \left(\frac{r}{r_c}\right)^{-1} + C$$

$r_c \equiv \frac{G M_\odot}{2 v_c^2}$: critical radius

$C = \text{constant}$
 $C = -3$ when $(r, v_r) = (r_c, v_c)$.

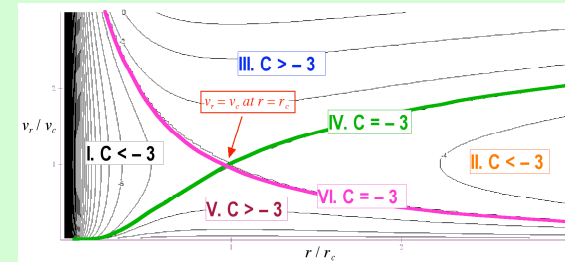


Characteristics of the solutions...

I. $C < -3$

II. $C < -3$

These two solutions are **physically incorrect**
(although they are correct mathematically)
because flow velocity takes two different values at the same position.



III. $C > -3$

IV. $C = -3$

V. $C > -3$

VI. $C = -3$

These four solutions are physically correct. Which is appropriate for solar wind?

III. $C > -3$

VI. $C = -3$

... starts with a **supersonic flow** ($v_r > v_c$) at a solar surface
=> inconsistent with inner boundary condition

IV. $C = -3$

... starts with a **subsonic flow** ($v_r < v_c$) at a solar surface, and becomes **supersonic**
=> consistent with inner boundary condition

V. $C > -3$

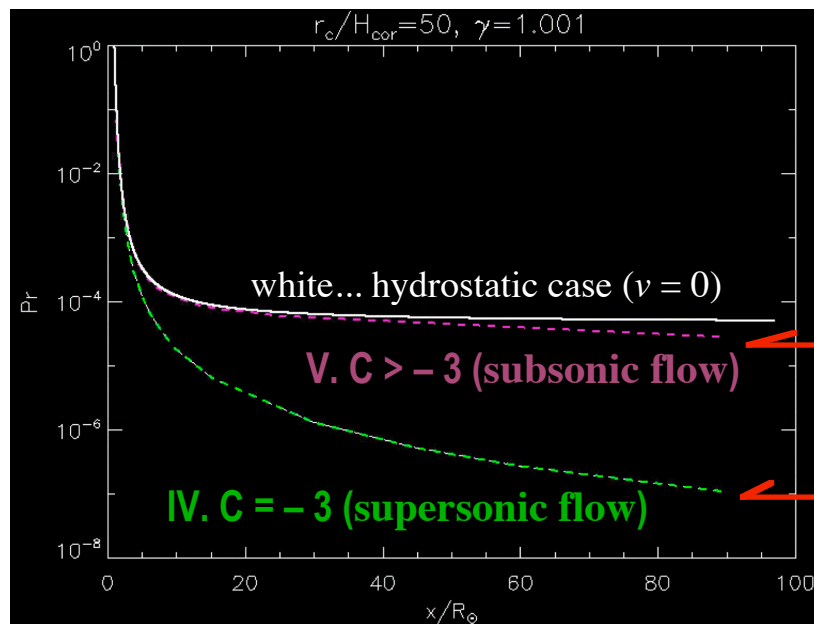
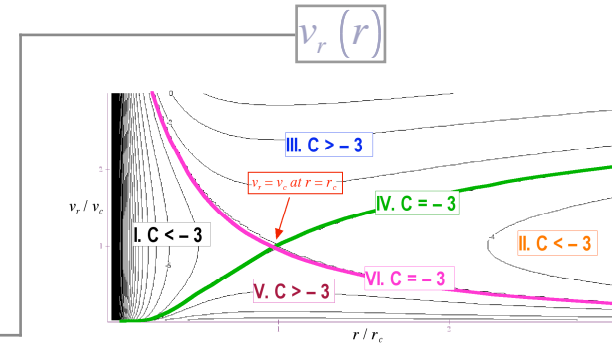
... starts with a **subsonic flow** ($v_r < v_c$) at a solar surface, and remains **subsonic**
=> consistent with inner boundary condition

Outer boundary condition (solution IV vs. solution V)...

We focus on gas pressure distribution:

$$p(r) = \rho(r) \frac{k_B T}{\bar{m}} = \rho(r) v_c^2, \quad (v_c \text{ is constant})$$

$$\rho(r) = \frac{\rho(R_\odot) v_r(R_\odot) R_\odot^2}{v_r(r) r^2}$$



$$r_c \sim 3 R_\odot \text{ with } T \sim 1.5 \times 10^6 \text{ K}$$

The value of gas pressure at outer boundary is different between IV (supersonic case, low pressure) and V (subsonic case, high pressure).

Observation suggests a very low value of gas pressure at the outer boundary ($P^{\text{interstellar}} \sim 10^{-17} P^{\text{solar surface}}$), so we select solution IV (supersonic case).